



Coatings for Space Flight

Composite Reflectors:

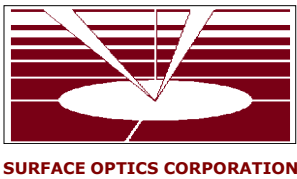
Reflectance

RF Conductivity

Thermal Control

July, 2012





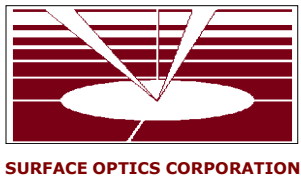
Overview

- **COMPANY:** Employee-owned San Diego company with 20 years experience in design, development, and application of precision coatings on critical spaceflight hardware.
- **COATINGS:** Ion Assisted Deposition (IAD) of metallic and dielectric films. Unique coating chambers and configurations to accommodate large substrates. Scalable processes.
- **CHAMBERS:** Four chambers accommodate a variety of applications. Reflectors and optics up to 3 METERS can be coated. Coating processes scalable to meet larger reflector needs.
- **FACILITIES:** 2,000 sq.ft. cleanroom (ISO Class 6) for cleaning, handling, coating. 100's of flight reflectors safely coated. Full design, fab, process capability for accommodating unique hardware.
- **VERIFICATION:** Internationally recognized expertise in measure of reflectance (200nm - 26microns), thermo-optical (α , ϵ), and surface scatter properties (BRDF).
- **CUSTOMERS:** NASA, AF, MDA, Axsys, Raytheon, L-3, Boeing, NG (Azusa, Baltimore, Xinetics), LM, ATK, Ball.
- **PROGRAMS:** DMSP, W-MAP, TDRS, OSTM, AEHF, Kepler, AMD

**1.5m Kepler Primary,
Coated w. Protected Ag,
for Ball/NASA**



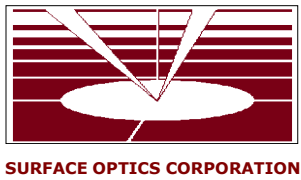
SOC Chamber for Parts up to 3 meters



SURFACE OPTICS CORPORATION



- *Optical Coatings*
- *Antenna and Spacecraft Coatings*
- *Optical Measurements*
- *Hyper-spectral Sensors*
- *Instruments*
- *Signature/Thermal Control*
- *Small Business in San Diego*
- *Top Secret Facility*
- *DOD and Commercial*



SOC BUSINESS UNITS

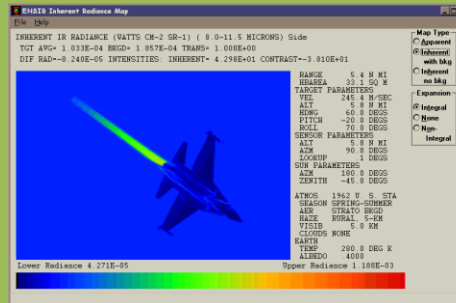
Coatings



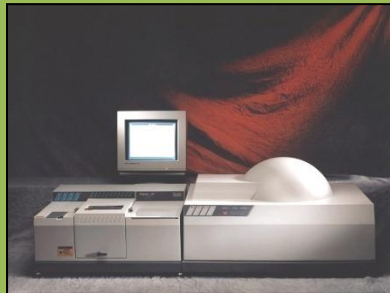
Hyper-spectral Sensors



Analysis



Instrumentation



Optical Measurements





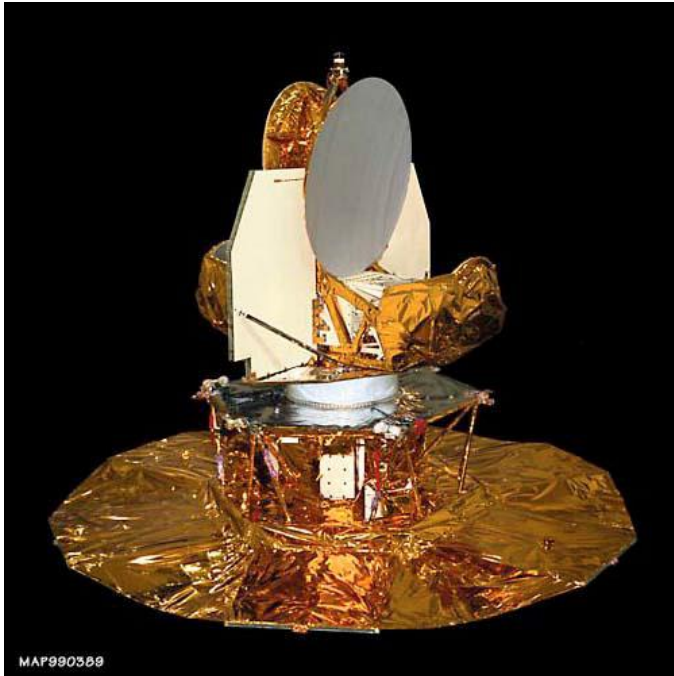
SOC Space Flight Heritage

Aluminum, Protected Aluminum, & Thermal Control Coatings

Program	Customer	Year	Reflector Size (m)	Thermal Requirements a/e	Radiometer	RF Conductivity Measured?
MLS	ATK (JPL)	1996	0.8 x 1.6	X		X
SeaWinds	JPL	1996	1.0	X		
Korea SAT		1997	1.0	X		
WMAP	GSFC	1998	1.5	X		
TDRS	BSS	1998	0.5			
BEAST Radio Telescope	ATK (UCSB)	1998	1.0, 2.2			
SSMIS	Aerojet (now NG)	1999	0.6	X	X	X
Jason 1	JPL	2000	0.5	X	X	
CMIS	ATK (Ball)	2000	2.3 x 2.5			
Cloudsat	ATK (JPL)	2002	1.8			X
WindSat	ATK (NRL)	2002	1.8	X	X	
DirectTV	ATK (LM)	2003	1.1			X
OSTM (Jason 2)	ATK (JPL)	2006	0.5	X	X	X
GMI	AASC (Ball)	2009	1.6			
AEHF	NGST	2009 - 2012	0.8 - 1.6			X
AMD	NG-Xinetics	2010 - 2012	1.2			
TDRS	AASC (BSS)	2011	0.5			
Jason 3	ATK (JPL)	2012	0.5	X	X	X



Early Developments



1.5 m WMAP Reflectors, GSFC



1.0 m Jason-1 Radiometer, NASA/JPL



2.2 m BEAST Telescope, UCSB

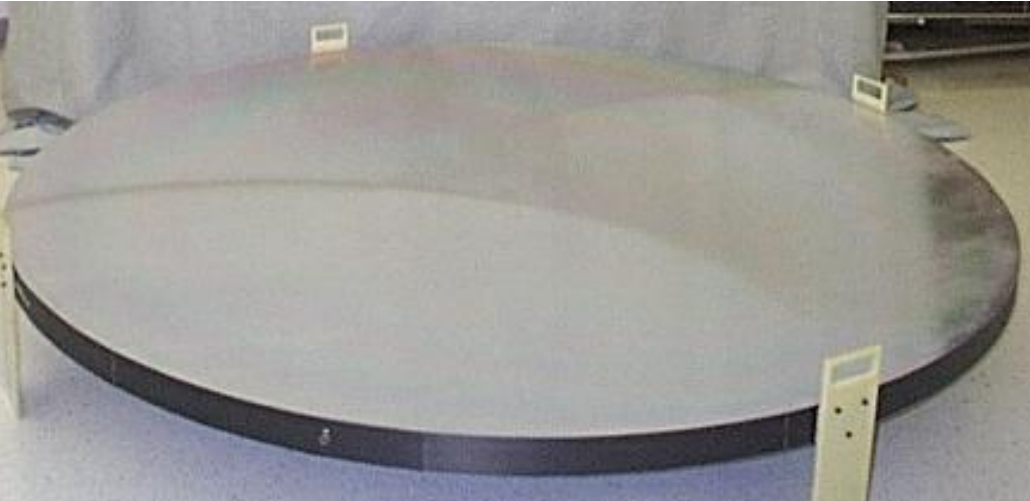
Large Apertures



1.8 m CloudSat Reflector, NASA/JPL



1.8 m WindSat Reflector, NRL

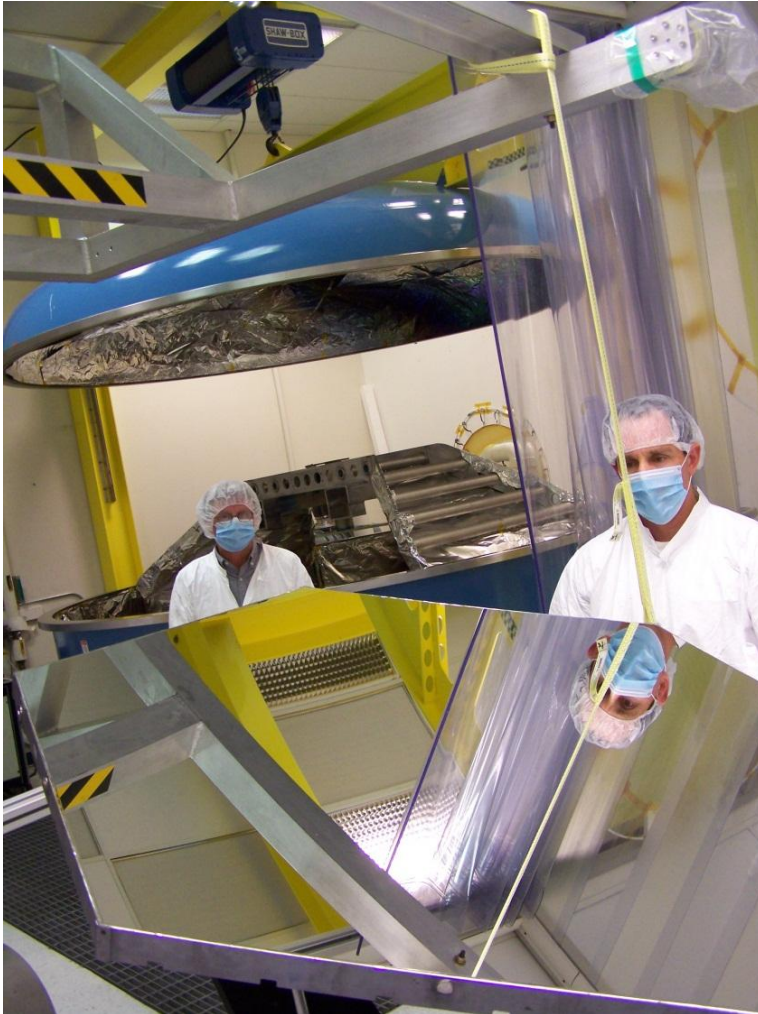


2.5 m CMIS Reflector, Ball



1.6 m AEHF Reflector, NGST

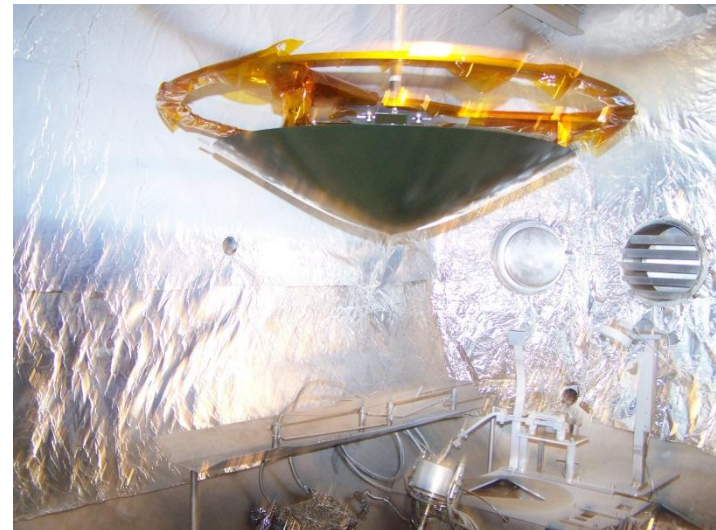
Recent Projects



1.2 m AMD Mirror, NG-Xinetics



1.0 m Jason-3 Radiometer, JPL



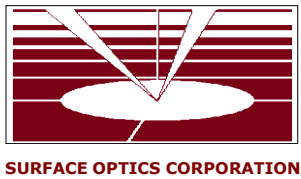
0.5 m TDRS Reflectors, Boeing



Chambers & Capabilities

- **3.3 meter (motion controlled e-beam IAD)**
- **1.8 meter (motion controlled e-beam IAD)**
- **1.2 meter (planetary)**
- **Small R&D chamber (up to 2 runs per day)**



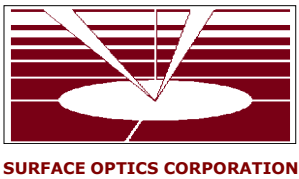


Key Coating Technologies

DESCRIPTION	APPLICATIONS	Technology Status
Ion Assisted E-beam Evaporation	Reflective & Transmissive coatings: optical, IR, RF	Industry standard
Moving Source Coating Platform	Enables coating of parts as large as chamber	Proprietary (SBIR data rights)
Aluminum coating w/ tailored thermal control properties	RF coating for composite spaceflight reflectors	Designated supplier in NG and BSS Specs
Low temperature coating of polymer substrates	Plastic lens, Nano-laminate Structures	Proprietary (SBIR data rights)

For more information, or a quote, contact: Mark Wesley, mwesley@surfaceoptics.com / 858-675-7404

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Substrate Handling Background

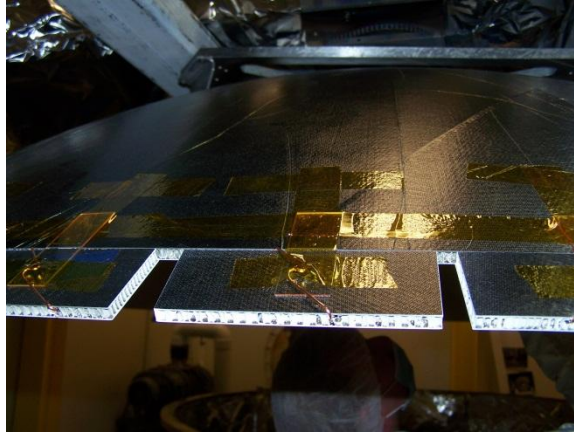
- SOC has safely processed hundreds of flight reflectors through our Coating facility.
- We have a full capability to address unique hardware
 - mechanical design
 - Fabrication
 - proof-loading
 - complete handling plans.
- Examples of some of our most complex Programs: Kepler, AMD, NIF, WMAP, Chandra, NuStar



AEHF 1.6m Composite Reflector



Unpacking and inspection



Edge coupons installed



Final inspection



Reflector installed in chamber, ready for coating



Surface Optics Quality System

Surface Optics quality system has been fully documented and implemented and is maintained as needed to meet the requirements of our Company vision and governing policies. Surface Optics has adopted a process-oriented method of management. This approach emphasizes the importance of continuous improvement and understanding, meeting and integrating customer requirements:



Page 1 of 1

This approval is subject to the company maintaining its system to the required standard, which will be monitored by NQA, USA, an accredited organization under the Aerospace Registrar Management Program.



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SOC Laboratory Measurement Expertise



BRDF
Emittance
Coating Thickness
Directional Reflectance
Hyper-spectral Discrimination
Surface Contamination
Optical Density
Coating Stress
Resistivity
Adhesion




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Optics Corporation

RF Conductivity

- Method developed by Aluizio Prata (USC) and JPL for 32GHz measurements.
- SOC now contracts directly with Aluizio Prata (California Electromagnetic Works) for these measurements.



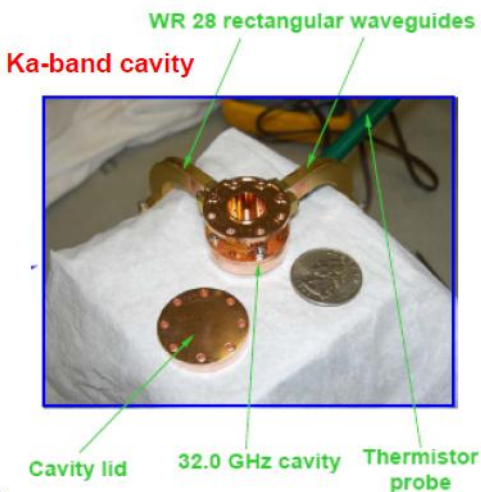
Resonant Cavity Technique



- **Resonant cavity technique yields effective conductivity of flat or gently curved surfaces**
- **Conductivity of cavity walls and lids first calibrated by measuring cavity Q at two resonant frequencies**
- **Top lid is replaced by material under test and Q at lower resonant frequency used to measure its effective conductivity, σ_E**

$$|\Gamma_H|^2 \equiv \left| \frac{\cos \theta_i - \frac{\eta_1}{\eta_2}}{\cos \theta_i + \frac{\eta_1}{\eta_2}} \right|^2, \quad |\Gamma_V|^2 \equiv \left| \frac{-\cos \theta_i + \frac{\eta_2}{\eta_1}}{\cos \theta_i + \frac{\eta_2}{\eta_1}} \right|^2,$$

$$\frac{\eta_1}{\eta_2} \equiv \sqrt{\frac{\sigma_E}{j2\pi f \epsilon_0}}.$$



WR 28 rectangular waveguides

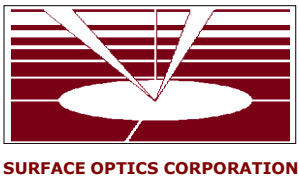
Ka-band cavity

Cavity lid

32.0 GHz cavity

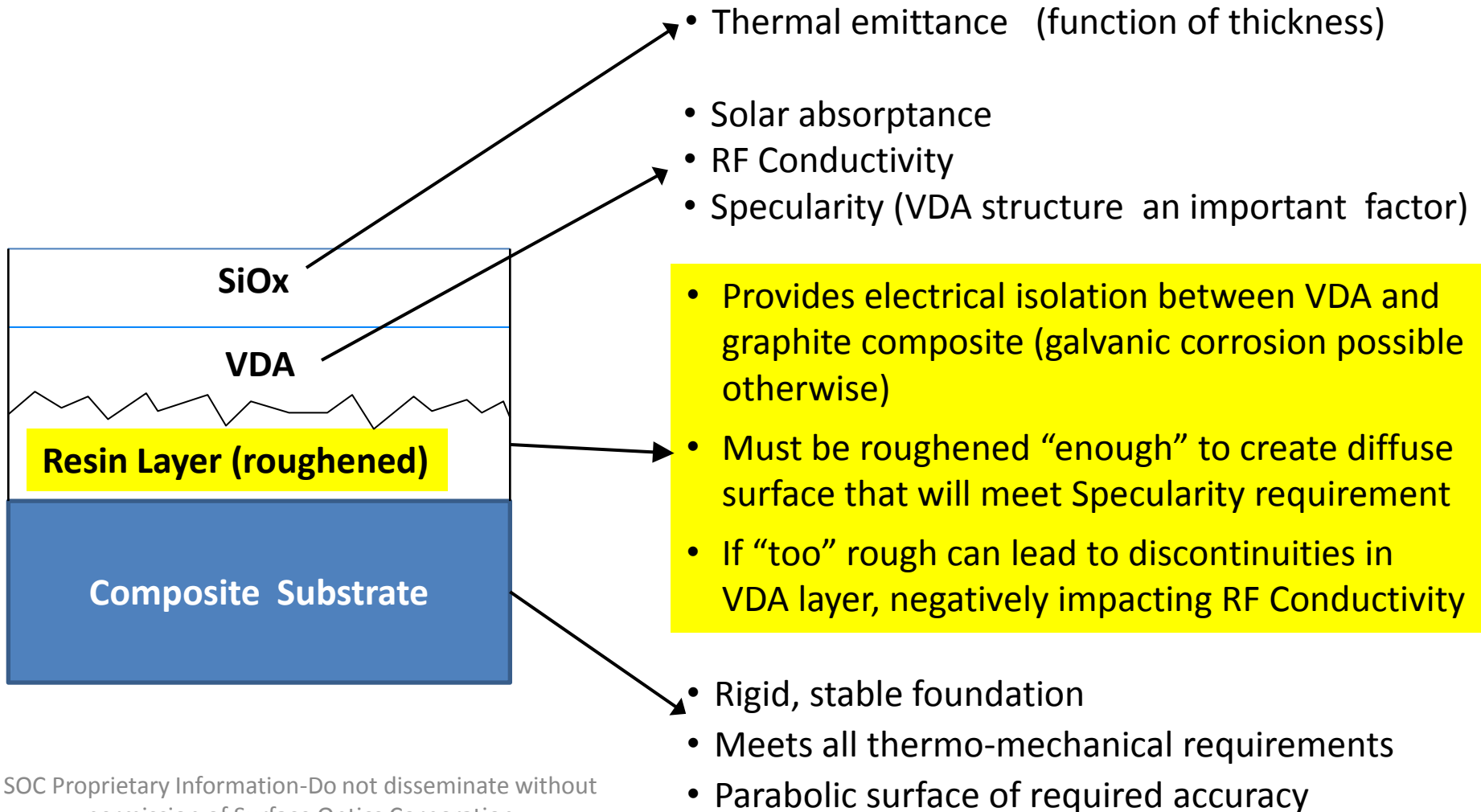
Thermistor probe

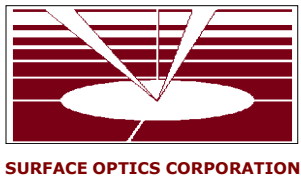
Brown et al. MicroR:



Thermal Control Coating Basics

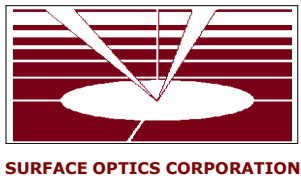
The design and application of a thermal control coating requires the balance of several competing requirements. Both Substrate and Coating are important.





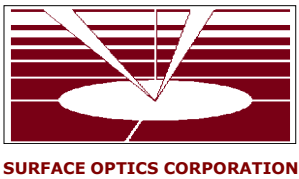
Thermal Control Coating Steps

- Composite substrate preparation is important to meet Specularity requirement, and mitigate galvanic potential with Aluminum
 - Epoxy resin washcoat, thermally condition, Gritblast
 - OR, apply primer in thin layers with matte finish (BR127 typical)
 - Measure resin surface prior to coating:
 - Measure Specularity on reflector surface using SOC handheld spectrometer
 - Measure DC resistivity to ensure electrical isolation
- Apply vapor deposited aluminum (VDA) with low solar absorptivity (α), in correct thickness for lowest op. frequency
- Apply overcoat (SiO_x or SiO_2) in correct thickness to yield desired value of emissivity (ϵ)
- Measure RF Conductivity, α , ϵ , and Specularity on both witness coupons, and coated reflector (if desired)



Typical VDA Requirements

- Reflective Coating
 - Aluminum Minimum Thickness
 - Reflectance
 - Adhesion
- Advanced Reflector Coating
 - DC Conductivity
 - RF Conductivity
 - Thermal Cycle
- Thermal Control Coating
 - Solar Absorptance (α)
 - IR Emittance (ε)
 - α/ε Ratio
 - Specularity (Solar Concentration)



Importance of VDA Process Parameters

- VDA process parameters determine aluminum microstructure.
- Microstructure influences competing coating performance properties:
 - Alpha
 - RF Conductivity
 - Specularity
 - Terrestrial stability (i.e. ground storage)
- Significant investment at SOC over last 10 years to understand, control, and document correct VDA processes

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Radiometer Coating Programs

Sample Summary

Project	Customer	Year	Size	Freq (GHz)	VDA	Overcoat	a/e	Diffusivity	Thermal Cycle	32 GHz RF Conductivity (MS/m)	Status
MAP	NASA	1998	0.6m	22 - 90	2.5 μ	2.4 μ SiOx	<<1	<5% , 10° cone			12 years on-orbit operation
WindSat	NRL	2002	1.8m	6 - 37							9 years on-orbit operation
MLS	ATK (COI)	2004	1.5m	63 - 205	1.2 μ	1.5 μ SiOx	1 - 2	<2% , 5° cone			8 years on-orbit operation
CloudSat	NASA/JPL	2005	2.5m	94	1.2 μ	1.1 μ SiOx	1 - 2	<5% , 10° cone			6 years on-orbit operation
OSTM	NASA/JPL	2008	1m	1 - 100	1.2 μ	1.1 μ SiOx	1 - 2	<25% , 10° cone	-120C / +135C		4 years on-orbit operation
GMI	NASA/ GSFC	2009	1.2m	10 - 183	1.7 μ	1.2 μ SiO ₂	< 1	< 0.4% , 1° cone	-100C / +150C		Flight article coated at JDSU. Thermal cycle reduced to +130C.
Jason 3	NASA / JPL	2012	1m	18 - 34	1.6 μ	1.1 μ SiOx	1 - 2	<25% , 10° cone	-120C / +135C		To Be Launched 2013

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